

## University of Anbar

### Engineering Thermodynamics II

Lecture Note #06: Mass and Energy Balances for a Steady-Flow Process

Mass and Energy balances for a steady-flow process

$$\sum_{in} \dot{m} = \sum_{out} \dot{m} \quad (\text{kg/s})$$

$$\dot{m}_1 = \dot{m}_2$$

Mass Balance

$$\rho_1 V_1 A_1 = \rho_2 V_2 A_2$$

A water heater in steady operation.

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\frac{dE_{system}}{dt}}_{\text{Rate of change in internal, kinetic, potential, etc., energies}} \xrightarrow{0 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in}}_{\text{Rate of net energy transfer in by heat, work, and mass}} = \underbrace{\dot{E}_{out}}_{\text{Rate of net energy transfer out by heat, work, and mass}} \quad (\text{kW})$$

$$\dot{Q}_{in} + \dot{W}_{in} + \sum_{in} \dot{m} \left( h + \frac{V^2}{2} + gz \right) = \dot{Q}_{out} + \dot{W}_{out} + \sum_{out} \dot{m} \left( h + \frac{V^2}{2} + gz \right)$$

for each inlet for each exit

### Energy balance relations with sign conventions (i.e., heat input and work output are positive)

$$\dot{Q} - \dot{W} = \sum_{\text{out}} \dot{m} \left( h + \frac{V^2}{2} + gz \right) - \sum_{\text{in}} \dot{m} \left( h + \frac{V^2}{2} + gz \right)$$

$$\dot{Q} - \dot{W} = \dot{m} \left[ h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} + g(z_2 - z_1) \right]$$

Where,  $q = \dot{Q}/\dot{m}$     $w = \dot{W}/\dot{m}$

$$q - w = h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} + g(z_2 - z_1)$$

when kinetic and potential energy changes are negligible

$$q - w = h_2 - h_1$$

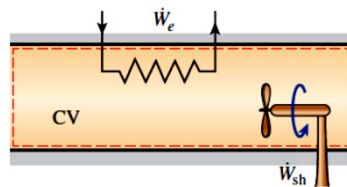


Fig. 5.22: Under steady operation, shaft work and electrical work are the only forms of work a simple compressible system may involve.

$$\frac{\text{J}}{\text{kg}} = \frac{\text{N} \cdot \text{m}}{\text{kg}} = \left( \frac{\text{kg} \cdot \text{m}}{\text{s}^2} \right) \frac{\text{m}}{\text{kg}} = \frac{\text{m}^2}{\text{s}^2}$$

(Also,  $\frac{\text{Btu}}{\text{lbm}} = 25,037 \frac{\text{ft}^2}{\text{s}^2}$ )

Fig 5.23: The  $\text{m}^2/\text{s}^2$  and  $\text{J}/\text{Kg}$  are equivalents

### Significant Remarks about Energy Balance Equation for Steady Flow

$\dot{Q}$  = **rate of heat transfer between the control volume and its surroundings**. When the control volume is losing heat (as in the case of the water heater),  $\dot{Q}$  is negative. If the control volume is well insulated (i.e., adiabatic), then  $\dot{Q} = 0$ .

$\dot{W}$  = **power**. For steady-flow devices, the control volume is constant; thus, there is no boundary work involved. The work required to push mass into and out of the control volume is also taken care of by using enthalpies for the energy of fluid streams instead of internal energies. Then  $\dot{W}$  represents the remaining forms of work done per unit time (Fig. 5–22). Many steady-flow devices, such as turbines, compressors, and pumps, transmit power through a shaft, and  $\dot{W}$  simply becomes the shaft power for those devices. If the control surface is crossed by electric wires (as in the case of an electric water heater),  $\dot{W}$  represents the electrical work done per unit time. If neither is present, then  $\dot{W} = 0$ .

$\Delta h = h_2 - h_1$ . The enthalpy change of a fluid can easily be determined by reading the enthalpy values at the exit and inlet states from the tables. For ideal gases, it can be approximated by  $\Delta h = c_{p,\text{avg}}(T_2 - T_1)$ . Note that  $(\text{kg/s})(\text{kJ/kg}) \equiv \text{kW}$ .

### Significant Remarks about Energy Balance Equation for Steady Flow

$\Delta ke = (V_2^2 - V_1^2)/2$ . The unit of kinetic energy is  $m^2/s^2$ , which is equivalent to J/kg (Fig. 5-23). The enthalpy is usually given in kJ/kg. To add these two quantities, the kinetic energy should be expressed in kJ/kg. This is easily accomplished by dividing it by 1000. A velocity of 45 m/s corresponds to a kinetic energy of only 1 kJ/kg, which is a very small value compared with the enthalpy values encountered in practice. Thus, the kinetic energy term at low velocities can be neglected. When a fluid stream enters and leaves a steady-flow device at about the same velocity ( $V_1 \cong V_2$ ), the change in the kinetic energy is close to zero regardless of the velocity. Caution should be exercised at high velocities, however, since small changes in velocities may cause significant changes in kinetic energy (Fig. 5-24).

|   | $V_1$ | $V_2$ | $\Delta ke$ |
|---|-------|-------|-------------|
|   | m/s   | m/s   | kJ/kg       |
| At very high velocities, even small changes in velocities can cause significant changes | 0     | 45    | 1           |
|   | 50    | 67    | 1           |
|   | 100   | 110   | 1           |
|   | 200   | 205   | 1           |
|   | 500   | 502   | 1           |

### Significant Remarks about Energy Balance Equation for Steady Flow

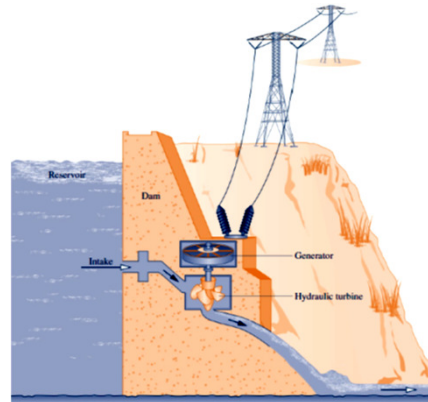
$\Delta pe = g(z_2 - z_1)$ . A similar argument can be given for the potential energy term. A potential energy change of 1 kJ/kg corresponds to an elevation difference of 102 m. The elevation difference between the inlet and exit of most industrial devices such as turbines and compressors is well below this value, and the potential energy term is always neglected for these devices. The only time the potential energy term is significant is when a process involves pumping a fluid to high elevations and we are interested in the required pumping power.



## Turbines

**Turbine** drives the electric generator in steam, gas, or hydroelectric power plants.

As the fluid passes through the turbine, work is done against the blades, which are attached to the shaft. As a result, the shaft rotates, and the turbine produces work.



$$\dot{Q}_{in} + \dot{W}_{in} + \sum_{in} \dot{m} \left( h + \frac{V^2}{2} + gz \right) = \dot{Q}_{out} + \dot{W}_{out} + \sum_{out} \dot{m} \left( h + \frac{V^2}{2} + gz \right)$$

for each inlet for each exit

Energy balance for the turbine in this figure:

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m} \left( h_1 + \frac{V_1^2}{2} + gz_1 \right) = \dot{W}_{out} + \dot{m} \left( h_2 + \frac{V_2^2}{2} + gz_2 \right)$$

(since  $\dot{Q} \cong \Delta pe \cong 0$ )

$$\dot{m}(h_1) = \dot{W}_{out} + \dot{m}(h_2)$$



Turbine blades attached to the turbine shaft

## Compressors

**Compressors**, as well as **pumps** and **fans**, are devices used to increase the pressure of a **fluid**. Work is supplied to these devices from an external source through a rotating shaft.

- A **fan** increases the pressure of a gas slightly and is mainly used to mobilize a gas.
- A **compressor** is capable of compressing the gas (vapor) to very high pressures.
- **Pumps** work very much like compressors except that they handle liquids instead of gases.

$$\dot{Q}_{in} + \dot{W}_{in} + \sum_{in} \dot{m} \left( h + \frac{V^2}{2} + gz \right) = \dot{Q}_{out} + \dot{W}_{out} + \sum_{out} \dot{m} \left( h + \frac{V^2}{2} + gz \right)$$

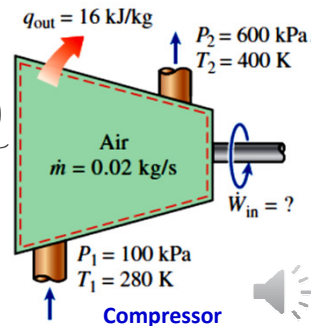
for each inlet for each exit

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{W}_{in} + \dot{m}h_1 = \dot{Q}_{out} + \dot{m}h_2$$

(since  $\Delta ke = \Delta pe \cong 0$ )

Energy balance for the compressor in this figure:



## General Notes on Turbines and Compressors

- **Turbines** produce power output ( $\dot{W}_{out}$ ) whereas **compressors, pumps, and fans** require power input ( $\dot{W}_{in}$ ).
- Heat transfer from turbines is usually negligible ( $\dot{Q} \approx 0$ ) since they are typically well insulated. Heat transfer is also negligible for compressors unless there is intentional cooling.
- Potential energy changes are negligible for all of these devices (Turbines, Compressors, Fans and Pumps) ( $\Delta Pe \approx 0$ ).
- The velocities involved in these devices, with the exception of turbines and fans, are usually too low to cause any significant change in the kinetic energy ( $\Delta Ke \approx 0$ ).
- The fluid velocities encountered in most turbines are very high, and the fluid experiences a significant change in its kinetic energy. However, this change is usually very small relative to the change in enthalpy, and thus it is often disregarded.



### EXAMPLE 5-6 Compressing Air by a Compressor

Air at 100 kPa and 280 K is compressed steadily to 600 kPa and 400 K. The mass flow rate of the air is 0.02 kg/s, and a heat loss of 16 kJ/kg occurs during the process. Assuming the changes in kinetic and potential energies are negligible, determine the necessary power input to the compressor.

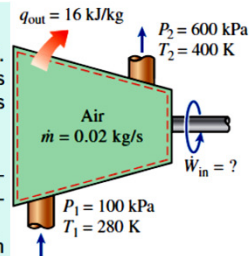
**SOLUTION** Air is compressed steadily by a compressor to a specified temperature and pressure. The power input to the compressor is to be determined.

**Assumptions** 1 This is a steady-flow process since there is no change with time at any point and thus  $\Delta m_{cv} = 0$  and  $\Delta E_{cv} = 0$ . 2 Air is an ideal gas since it is at a high temperature and low pressure relative to its critical-point values. 3 The kinetic and potential energy changes are zero,  $\Delta ke = \Delta pe = 0$ .

**Analysis** We take the *compressor* as the system (Fig. 5-30). This is a *control volume* since mass crosses the system boundary during the process. We observe that there is only one inlet and one exit and thus  $\dot{m}_1 = \dot{m}_2 = \dot{m}$ . Also, heat is lost from the system and work is supplied to the system.

Under stated assumptions and observations, the energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\frac{dE_{system}/dt}_{\text{Rate of change in internal, kinetic, potential, etc., energies}}}_{0 \text{ (steady)}} = 0$$





$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{W}_{\text{in}} + \dot{m}h_1 = \dot{Q}_{\text{out}} + \dot{m}h_2 \quad (\text{since } \Delta \text{ke} = \Delta \text{pe} \cong 0)$$

$$\dot{W}_{\text{in}} = \dot{m}q_{\text{out}} + \dot{m}(h_2 - h_1)$$

The enthalpy of an ideal gas depends on temperature only, and the enthalpies of the air at the specified temperatures are determined from the air table (Table A-17) to be

$$h_1 = h_{@ 280 \text{ K}} = 280.13 \text{ kJ/kg}$$

$$h_2 = h_{@ 400 \text{ K}} = 400.98 \text{ kJ/kg}$$

Substituting, the power input to the compressor is determined to be

$$\begin{aligned} \dot{W}_{\text{in}} &= (0.02 \text{ kg/s})(16 \text{ kJ/kg}) + (0.02 \text{ kg/s})(400.98 - 280.13) \text{ kJ/kg} \\ &= \mathbf{2.74 \text{ kW}} \end{aligned}$$

**Discussion** Note that the mechanical energy input to the compressor manifests itself as a rise in enthalpy of air and heat loss from the compressor.

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### EXAMPLE 5-7 Power Generation by a Steam Turbine

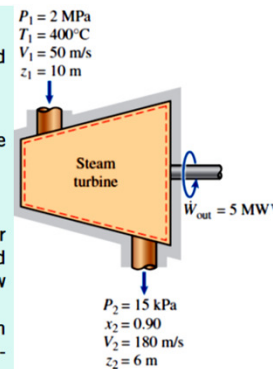
The power output of an adiabatic steam turbine is 5 MW, and the inlet and the exit conditions of the steam are as indicated in Fig. 5-31.

- Compare the magnitudes of  $\Delta h$ ,  $\Delta \text{ke}$ , and  $\Delta \text{pe}$ .
- Determine the work done per unit mass of the steam flowing through the turbine.
- Calculate the mass flow rate of the steam.

**SOLUTION** The inlet and exit conditions of a steam turbine and its power output are given. The changes in kinetic energy, potential energy, and enthalpy of steam, as well as the work done per unit mass and the mass flow rate of steam are to be determined.

**Assumptions** 1 This is a steady-flow process since there is no change with time at any point and thus  $\Delta m_{\text{CV}} = 0$  and  $\Delta E_{\text{CV}} = 0$ . 2 The system is adiabatic and thus there is no heat transfer.

**Analysis** We take the *turbine* as the system. This is a *control volume* since mass crosses the system boundary during the process. We observe that there is only one inlet and one exit and thus  $\dot{m}_1 = \dot{m}_2 = \dot{m}$ . Also, work is done by the system. The inlet and exit velocities and elevations are given, and thus the kinetic and potential energies are to be considered.



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(a) At the inlet, steam is in a superheated vapor state, and its enthalpy is

$$\left. \begin{array}{l} P_1 = 2 \text{ MPa} \\ T_1 = 400^\circ\text{C} \end{array} \right\} h_1 = 3248.4 \text{ kJ/kg} \quad (\text{Table A-6})$$

At the turbine exit, we obviously have a saturated liquid–vapor mixture at 15-kPa pressure. The enthalpy at this state is

$$h_2 = h_f + x_2 h_{fg} = [225.94 + (0.9)(2372.3)] \text{ kJ/kg} = 2361.01 \text{ kJ/kg}$$

Then

$$\Delta h = h_2 - h_1 = (2361.01 - 3248.4) \text{ kJ/kg} = -887.39 \text{ kJ/kg}$$

$$\Delta \text{ke} = \frac{V_2^2 - V_1^2}{2} = \frac{(180 \text{ m/s})^2 - (50 \text{ m/s})^2}{2} \left( \frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) = 14.95 \text{ kJ/kg}$$

$$\Delta \text{pe} = g(z_2 - z_1) = (9.81 \text{ m/s}^2)[(6 - 10) \text{ m}] \left( \frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) = -0.04 \text{ kJ/kg}$$

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(b) The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\frac{dE_{\text{system}}}{dt}}_{\text{Rate of change in internal, kinetic, potential, etc., energies}} \overset{0 \text{ (steady)}}{=} 0$$

$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{m} \left( h_1 + \frac{V_1^2}{2} + gz_1 \right) = \dot{W}_{\text{out}} + \dot{m} \left( h_2 + \frac{V_2^2}{2} + gz_2 \right) \quad (\text{since } \dot{Q} = 0)$$

Dividing by the mass flow rate  $\dot{m}$  and substituting, the work done by the turbine per unit mass of the steam is determined to be

$$w_{\text{out}} = - \left[ (h_2 - h_1) + \frac{V_2^2 - V_1^2}{2} + g(z_2 - z_1) \right] = -(\Delta h + \Delta \text{ke} + \Delta \text{pe})$$

$$= -[-887.39 + 14.95 - 0.04] \text{ kJ/kg} = 872.48 \text{ kJ/kg}$$

(c) The required mass flow rate for a 5-MW power output is

$$\dot{m} = \frac{\dot{W}_{\text{out}}}{w_{\text{out}}} = \frac{5000 \text{ kJ/s}}{872.48 \text{ kJ/kg}} = 5.73 \text{ kg/s}$$

**Discussion** Two observations can be made from these results. First, the change in potential energy is insignificant in comparison to the changes in enthalpy and kinetic energy. This is typical for most engineering devices. Second, as a result of low pressure and thus high specific volume, the steam velocity at the turbine exit can be very high. Yet the change in kinetic energy is a small fraction of the change in enthalpy (less than 2 percent in our case) and is therefore often neglected.

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**Example 3**

Air flows steadily through an adiabatic turbine, entering at 150 psia, 900°F, and 350 ft/s and leaving at 20 psia, 300°F, and 700 ft/s. The inlet area of the turbine is 0.1 ft<sup>2</sup>. Determine (a) the mass flow rate of the air and (b) the power output of the turbine.

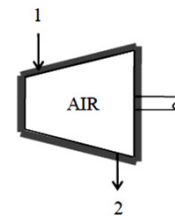
**Assumptions** 1 This is a steady-flow process since there is no change with time. 2 Potential energy changes are negligible. 3 The device is adiabatic and thus heat transfer is negligible. 4 Air is an ideal gas with constant specific heats.

**Properties** The gas constant of air is  $R = 0.3704 \text{ psia}\cdot\text{ft}^3/\text{lbm}\cdot\text{R}$ . The constant pressure specific heat of air at the average temperature of  $(900 + 300)/2 = 600^\circ\text{F}$  is  $c_p = 0.25 \text{ Btu}/\text{lbm}\cdot^\circ\text{F}$  (Table A-2a)

**Analysis** (a) There is only one inlet and one exit, and thus  $\dot{m}_1 = \dot{m}_2 = \dot{m}$ . The inlet specific volume of air and its mass flow rate are

$$v_1 = \frac{RT_1}{P_1} = \frac{(0.3704 \text{ psia}\cdot\text{ft}^3/\text{lbm}\cdot\text{R})(1360 \text{ R})}{150 \text{ psia}} = 3.358 \text{ ft}^3/\text{lbm}$$

$$\dot{m} = \frac{1}{v_1} A_1 V_1 = \frac{1}{3.358 \text{ ft}^3/\text{lbm}} (0.1 \text{ ft}^2)(350 \text{ ft/s}) = 10.42 \text{ lbm/s}$$



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**Example 3 Cont..**

(b) We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} \stackrel{\text{no (steady)}}{=} 0$$

$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{m}(h_1 + V_1^2/2) = \dot{W}_{\text{out}} + \dot{m}(h_2 + V_2^2/2) \quad (\text{since } \dot{Q} \equiv \Delta \text{pe} \equiv 0)$$

$$\dot{W}_{\text{out}} = -\dot{m} \left( h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} \right) = -\dot{m} \left( c_p(T_2 - T_1) + \frac{V_2^2 - V_1^2}{2} \right)$$

$$\begin{aligned} \dot{W}_{\text{out}} &= -(10.42 \text{ lbm/s}) \left[ (0.250 \text{ Btu}/\text{lbm}\cdot^\circ\text{F})(300 - 900)^\circ\text{F} + \frac{(700 \text{ ft/s})^2 - (350 \text{ ft/s})^2}{2} \left( \frac{1 \text{ Btu}/\text{lbm}}{25.037 \text{ ft}^2/\text{s}^2} \right) \right] \\ &= 1486.5 \text{ Btu/s} = \mathbf{1568 \text{ kW}} \end{aligned}$$

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**Example 4**

Refrigerant-134a enters an adiabatic compressor as saturated vapor at  $-24^{\circ}\text{C}$  and leaves at  $0.8\text{ MPa}$  and  $60^{\circ}\text{C}$ . The mass flow rate of the refrigerant is  $1.2\text{ kg/s}$ . Determine (a) the power input to the compressor and (b) the volume flow rate of the refrigerant at the compressor inlet.

**Assumptions** 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 The device is adiabatic and thus heat transfer is negligible.

**Properties** From the refrigerant tables (Tables A-11 through 13)

$$\left. \begin{array}{l} T_1 = -24^{\circ}\text{C} \\ \text{sat. vapor} \end{array} \right\} \begin{array}{l} v_1 = 0.17398\text{ m}^3/\text{kg} \\ h_1 = 235.94\text{ kJ/kg} \end{array} \quad \left. \begin{array}{l} P_2 = 0.8\text{ MPa} \\ T_2 = 60^{\circ}\text{C} \end{array} \right\} h_2 = 296.82\text{ kJ/kg}$$

**Analysis** (a) There is only one inlet and one exit, and thus  $\dot{m}_1 = \dot{m}_2 = \dot{m}$ . We take the compressor as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\substack{\text{Rate of net energy transfer} \\ \text{by heat, work, and mass}}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\substack{\text{Rate of change in internal, kinetic,} \\ \text{potential, etc. energies}}} \stackrel{\neq 0 \text{ (steady)}}{=} 0$$

$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{W}_{\text{in}} + \dot{m}h_1 = \dot{m}h_2 \quad (\text{since } \dot{Q} \equiv \Delta ke \equiv \Delta pe \equiv 0)$$

$$\dot{W}_{\text{in}} = \dot{m}(h_2 - h_1)$$

$$\dot{W}_{\text{in}} = (1.2\text{ kg/s})(296.82 - 235.94)\text{ kJ/kg} = \mathbf{73.06\text{ kJ/s}}$$

(b) The volume flow rate of the refrigerant at the compressor inlet is

$$\dot{V}_1 = \dot{m}v_1 = (1.2\text{ kg/s})(0.17398\text{ m}^3/\text{kg}) = \mathbf{0.209\text{ m}^3/\text{s}}$$

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**Example 5**

Refrigerant-134a enters a compressor at  $180\text{ kPa}$  as a saturated vapor with a flow rate of  $0.35\text{ m}^3/\text{min}$  and leaves at  $700\text{ kPa}$ . The power supplied to the refrigerant during compression process is  $2.35\text{ kW}$ . What is the temperature of R-134a at the exit of the compressor?

**Assumptions** 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat transfer with the surroundings is negligible.

**Analysis** We take the compressor as the system, which is a control volume since mass crosses the boundary. Noting that one fluid stream enters and leaves the compressor, the energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\substack{\text{Rate of net energy transfer} \\ \text{by heat, work, and mass}}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\substack{\text{Rate of change in internal, kinetic,} \\ \text{potential, etc. energies}}} \stackrel{\neq 0 \text{ (steady)}}{=} 0$$

$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{W}_{\text{in}} + \dot{m}h_1 = \dot{m}h_2 \quad (\text{since } \Delta ke \equiv \Delta pe \equiv 0)$$

$$\dot{W}_{\text{in}} = \dot{m}(h_2 - h_1)$$

From R134a tables (Table A-12)

$$\left. \begin{array}{l} P_1 = 180\text{ kPa} \\ x_1 = 0 \end{array} \right\} \begin{array}{l} h_1 = 242.90\text{ kJ/kg} \\ v_1 = 0.1105\text{ m}^3/\text{kg} \end{array}$$

The mass flow rate is

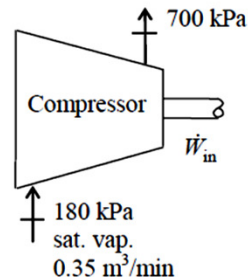
$$\dot{m} = \frac{\dot{V}_1}{v_1} = \frac{(0.35/60)\text{ m}^3/\text{s}}{0.1104\text{ m}^3/\text{kg}} = 0.05283\text{ kg/s}$$

Substituting for the exit enthalpy,

$$\dot{W}_{\text{in}} = \dot{m}(h_2 - h_1)$$

$$2.35\text{ kW} = (0.05283\text{ kg/s})(h_2 - 242.90)\text{ kJ/kg} \longrightarrow h_2 = 287.41\text{ kJ/kg}$$

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**Example 5      Cont..**

From Table A-13,

$$\left. \begin{array}{l} P_2 = 700 \text{ kPa} \\ h_2 = 287.41 \text{ kJ/kg} \end{array} \right\} T_2 = 48.9^\circ\text{C}$$

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**Example 6**

Steam flows steadily through an adiabatic turbine. The inlet conditions of the steam are 4 MPa, 500°C, and 80 m/s, and the exit conditions are 30 kPa, 92 percent quality, and 50 m/s. The mass flow rate of the steam is 12 kg/s. Determine (a) the change in kinetic energy, (b) the power output, and (c) the turbine inlet area.

*Assumptions* 1 This is a steady-flow process since there is no change with time. 2 Potential energy changes are negligible. 3 The device is adiabatic and thus heat transfer is negligible.

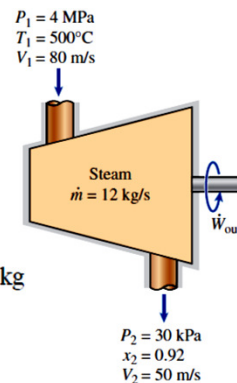
*Properties* From the steam tables (Tables A-4 through 6)

$$\left. \begin{array}{l} P_1 = 4 \text{ MPa} \\ T_1 = 500^\circ\text{C} \end{array} \right\} \begin{array}{l} v_1 = 0.086442 \text{ m}^3/\text{kg} \\ h_1 = 3446.0 \text{ kJ/kg} \end{array}$$

$$\left. \begin{array}{l} P_2 = 30 \text{ kPa} \\ x_2 = 0.92 \end{array} \right\} h_2 = h_f + x_2 h_{fg} = 289.27 + 0.92 \times 2335.3 = 2437.7 \text{ kJ/kg}$$

*Analysis* (a) The change in kinetic energy is determined from

$$\Delta ke = \frac{V_2^2 - V_1^2}{2} = \frac{(50 \text{ m/s})^2 - (80 \text{ m/s})^2}{2} \left( \frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) = -1.95 \text{ kJ/kg}$$



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**Example 6 Cont..**

(b) There is only one inlet and one exit, and thus  $\dot{m}_1 = \dot{m}_2 = \dot{m}$ . We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} \stackrel{\approx 0 \text{ (steady)}}{=} 0$$

$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{m}(h_1 + V_1^2/2) = \dot{W}_{\text{out}} + \dot{m}(h_2 + V_2^2/2) \quad (\text{since } \dot{Q} \equiv \Delta pe \equiv 0)$$

$$\dot{W}_{\text{out}} = -\dot{m} \left( h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} \right)$$

Then the power output of the turbine is determined by substitution to be

$$\dot{W}_{\text{out}} = -(12 \text{ kg/s})(2437.7 - 3446.0 - 1.95) \text{ kJ/kg} = 12,123 \text{ kW} = \mathbf{12.1 \text{ MW}}$$

(c) The inlet area of the turbine is determined from the mass flow rate relation,

$$\dot{m} = \frac{1}{v_1} A_1 V_1 \longrightarrow A_1 = \frac{\dot{m} v_1}{V_1} = \frac{(12 \text{ kg/s})(0.086442 \text{ m}^3/\text{kg})}{80 \text{ m/s}} = \mathbf{0.0130 \text{ m}^2}$$

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**Example 7**

Steam enters an adiabatic turbine at 10 MPa and 500°C and leaves at 10 kPa with a quality of 90 percent. Neglecting the changes in kinetic and potential energies, determine the mass flow rate required for a power output of 5 MW.

**Assumptions** 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 The device is adiabatic and thus heat transfer is negligible.

**Properties** From the steam tables (Tables A-4 through 6)

$$\left. \begin{array}{l} P_1 = 10 \text{ MPa} \\ T_1 = 500^\circ\text{C} \end{array} \right\} h_1 = 3375.1 \text{ kJ/kg}$$

$$\left. \begin{array}{l} P_2 = 10 \text{ kPa} \\ x_2 = 0.90 \end{array} \right\} h_2 = h_f + x_2 h_{fg} = 191.81 + 0.90 \times 2392.1 = 2344.7 \text{ kJ/kg}$$

**Analysis** There is only one inlet and one exit, and thus  $\dot{m}_1 = \dot{m}_2 = \dot{m}$ . We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} \stackrel{\approx 0 \text{ (steady)}}{=} 0$$

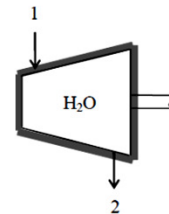
$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{m}h_1 = \dot{W}_{\text{out}} + \dot{m}h_2 \quad (\text{since } \dot{Q} \equiv \Delta ke \equiv \Delta pe \equiv 0)$$

$$\dot{W}_{\text{out}} = -\dot{m}(h_2 - h_1)$$

Substituting, the required mass flow rate of the steam is determined to be

$$5000 \text{ kJ/s} = -\dot{m}(2344.7 - 3375.1) \text{ kJ/kg} \longrightarrow \dot{m} = \mathbf{4.852 \text{ kg/s}}$$



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### Example 8

Steam flows steadily through a turbine at a rate of 45,000 lbm/h, entering at 1000 psia and 900°F and leaving at 5 psia as saturated vapor. If the power generated by the turbine is 4 MW, determine the rate of heat loss from the steam.

**Assumptions** 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible.

**Properties** From the steam tables (Tables A-4E through 6E)

$$\left. \begin{array}{l} P_1 = 1000 \text{ psia} \\ T_1 = 900^\circ\text{F} \end{array} \right\} h_1 = 1448.6 \text{ Btu/lbm}$$

$$\left. \begin{array}{l} P_2 = 5 \text{ psia} \\ \text{sat. vapor} \end{array} \right\} h_2 = 1130.7 \text{ Btu/lbm}$$

**Analysis** There is only one inlet and one exit, and thus  $\dot{m}_1 = \dot{m}_2 = \dot{m}$ . We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} \stackrel{\text{steady}}{=} 0$$

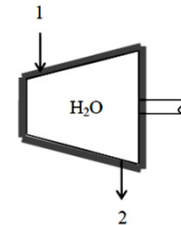
$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{m}h_1 = \dot{Q}_{\text{out}} + \dot{W}_{\text{out}} + \dot{m}h_2 \quad (\text{since } \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{Q}_{\text{out}} = -\dot{m}(h_2 - h_1) - \dot{W}_{\text{out}}$$

Substituting,

$$\dot{Q}_{\text{out}} = -(45000/3600 \text{ lbm/s})(1130.7 - 1448.6) \text{ Btu/lbm} - 4000 \text{ kJ/s} \left( \frac{1 \text{ Btu}}{1.055 \text{ kJ}} \right) = \mathbf{182.0 \text{ Btu/s}}$$



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### Example 9

Helium is to be compressed from 105 kPa and 295 K to 700 kPa and 460 K. A heat loss of 15 kJ/kg occurs during the compression process. Neglecting kinetic energy changes, determine the power input required for a mass flow rate of 60 kg/min.

**Assumptions** 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Helium is an ideal gas with constant specific heats.

**Properties** The constant pressure specific heat of helium is  $c_p = 5.1926 \text{ kJ/kg} \cdot \text{K}$  (Table A-2a).

**Analysis** There is only one inlet and one exit, and thus  $\dot{m}_1 = \dot{m}_2 = \dot{m}$ . We take the compressor as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} \stackrel{\text{steady}}{=} 0$$

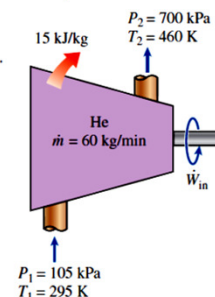
$$\dot{E}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{W}_{\text{in}} + \dot{m}h_1 = \dot{Q}_{\text{out}} + \dot{m}h_2 \quad (\text{since } \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{W}_{\text{in}} - \dot{Q}_{\text{out}} = \dot{m}(h_2 - h_1) = \dot{m}c_p(T_2 - T_1)$$

Thus,

$$\begin{aligned} \dot{W}_{\text{in}} &= \dot{Q}_{\text{out}} + \dot{m}c_p(T_2 - T_1) \\ &= (60/60 \text{ kg/s})(15 \text{ kJ/kg}) + (60/60 \text{ kg/s})(5.1926 \text{ kJ/kg} \cdot \text{K})(460 - 295) \text{ K} \\ &= \mathbf{872 \text{ kW}} \end{aligned}$$



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